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Surface & Coatings Technology xx (2004) xxx-xxx



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Toughness measurement of thin films: a critical review

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6 Abstract

At present, there is neither standard test procedure nor standard methodology for assessment of toughness of thin films. However, researchers have long been trying to make such measurements, thus a spectrum of test methods have been developed, mostly each in its own way. As qualitative or semiquantitative assessment, a simple plasticity measurement or scratch adhesion test can mostly suffice. For quantitative description, however, a choice of bending, buckling, indentation, scratching, or tensile test has to be made. These testing methods are either stress-based or energy-based. This paper gives a critical review on these methods and concludes that, for thin films, the energybased approach, especially the one independent of substrate, is more advantageous.

13 © 2004 Published by Elsevier B.V.

14 PACS: 68.55.-a; 68.90.+g; 68.35.Gy

15 Keywords: Toughness; Toughness measurement; Thin films; Coatings

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17 1. Introduction

18In essence, toughness is the ability of a material to absorb 19energy during deformation up to fracture [1,2]. Fracture 20toughness is the ability of a material to resist the growth of a preexisting crack. According to this definition, toughness 2122encompasses the energy required both to create the crack 23and to enable the crack to propagate until fracture, whereas fracture toughness takes only account of the energy required 24to facilitate the crack propagation to fracture. These are two 25different concepts and should not be confused and 26interchangeably used. For bulk materials and some thick 2728films, fracture toughness is easily measured according to ASTM standards [3,4]. However, for thin films, fracture 2930 toughness measurement remains difficult because of the 31 thickness limitation [5]. As thin films are increasingly 32 finding their way in engineering applications, thin film 33 toughness assessment becomes imperative. Unlike the bulk materials, however, until now, there is neither standard 34procedure nor commonly accepted methodology to follow. 3536 It is, however, good to note that increasing efforts have been 37 made to address this tricky issue, and thus quite many test

methods are proposed and used in various literatures. This38paper attempts to size up these methods, compare, and sort39out the critical issues. An effort is made to confine the scope40to hard and superhard thin films (thus, soft thin films fall out41of the scope). To avoid unnecessary complication, the word42"films" is used in this paper to mean films as well as43coatings.44

2. Various toughness methodologies

The methodologies employed to measure toughness for 46 thin films fall into one of these methods: bending, buckling, 47 adhesion scratch test, indentation, and tensile tests, which 48 are discussed in details below. 49

2.1. Bending 50

For freestanding thick films of tens or hundreds of 51 microns in thickness, measurement of the fracture toughness can be very similar to that for a bulk material: 53 creating a precrack, applying a stress to induce crack 54 propagation, and then measuring the critical stress needed 55 to inflict fracture. However, introduction of a precrack of 56 known size in a film, especially in a thin film of only 57

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 $^{0257\}text{-}8972/\$$ - see front matter C 2004 Published by Elsevier B.V. doi:10.1016/j.surfcoat.2004.10.021

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micron size or submicron size thick, is extremely tricky. 5859In Ref. [6], a freestanding diamond film with thickness in 60 the order of millimeters was laser-cut at one edge to form a notch and then glued onto the side face of a brass 61 62beam (cf. Fig. 1). The brass beam was bent so that a precrack was generated in the film at the end of the 63 notch. The film was then removed from the beam and 64put under a three-point flexure to bend, as illustrated in 6566 Fig. 2.

The fracture toughness was thus calculated according toASTM standard E-399 [3] using

$$K_{Ic} = \left(P_{\rm c}S/hW^{2/3}\right)f(a/W) \tag{1}$$

60 where $P_{\rm c}$ is the load at fracture, h and W are the thickness 71 and the width of the film, respectively, S is the span between 72 the two supporting positions, and a is the length of the preexisting crack, f is a function of a/W. That is not an easy 73 74 experiment, let alone the possible errors easily introduced 75 during gluing and ungluing, nonsymmetrical propagation of 76 crack in the brass plates, etc. Obviously, this method is not 77 applicable for thin films.

78Jaeger et al. [7] used an ingenious way to make the 79precrack: a notch was first made on the front face of the 80 steel substrate, and a hole was bored at the end (cf. Fig. 3). 81 The substrate was then fatigued to generate a crack from the notch that propagated and stopped at the hole. After that, a 82 83 few-micron-thick film was deposited onto the side faces of 84 the substrate. The coated substrate then underwent two successive four-point bending tests: one in the as-coated 85 86 state, the other after the film was broken. During the 87 bending, the load was recorded as a function of the 88 displacement. When the crack reached the hole, the load 89 was removed and then replaced for the second flexure. The 90 second bending was stopped as the displacement reached the value of the first bending. The difference in load F was 91 92considered the load required for crack propagation in the 93 film. During the two successive bending processes, if the 94substrate remained elastic, the energy difference (ΔU_{e}) 95would represent the energy required to enable cracking of



Fig. 2. Schematic diagram of three-point bending test of a freestanding diamond films with precrack [6].

the film. The critical energy release rate G_c can be written 96 as 97

$$G_{\rm c} = -\frac{\mathrm{d}U_{\rm e}}{\mathrm{d}A} = -\frac{-\frac{1}{2}F^2 \cdot \mathrm{d}C}{2h \cdot \mathrm{d}a} \tag{2}$$

where U_e is the elastic energy, A is the area of the crack, h 98 is the film thickness, dC/da is the change in the compliance 100 (C) of the film with respect to the change in crack length 101 (a), F is the measured force difference in the two 102 successive bending. Under the plain stress condition (that 103 is, the film thickness is significantly less than length and 104 width), fracture toughness K_C can be calculated from G_c 105 through 106

$$K_c = \sqrt{EG_c} \tag{3}$$

Testing of TiN, TiCN, and TiAlN films of 4.8 to 7.9 µm 109 thick [deposited via plasma-assisted chemical vapor 110 deposition (PACVD)] yields a fracture toughness of these 111 films as 8.7, 7.9, and 3.8 MPa $m^{1/2}$, respectively. However, 112as the authors readily pointed out, the introduction of 113precrack into substrate may have significant effects on film 114formation and growth during deposition process, therefore 115affecting the fracture toughness values. In addition, should 116there be any plastic deformation in the substrate, energy 117



Fig. 1. Schematic diagram of introducing precrack in film using bending method [6].



Fig. 3. Substrate with precrack for hard thin film toughness measurement via four-point bending; the film will be deposited on the side face.

118 measurement would be wrong, and thus affecting the 119 toughness calculation.

120Preparation of precise precrack in films is usually difficult and inconvenient. This is especially true for hard 121films with thickness of microns or submicrons. Therefore, 122123 bending without precrack is adopted by some researchers 124 [8-10], and "cracking resistance" is indirectly used as a 125 measure of fracture toughness. The cracking resistance is defined as the threshold strain over which density of the 126crack sharply increases. The onset of the increase of the 127 128 number of cracks is detectable by significant increase in 129 acoustic emission from a detector mounted on the film 130 [11–13] or by directly measuring the crack density as a 131 function of strain [14,15]. Installing the flexure in a 132 scanning electron microscope (SEM) facilitates "live" 133 monitoring of the formation of cracks [16]. Wiklund et 134 al. [17] measured the cracking resistance of CrN (2.5 µm 135 thick) and TiN (4.3 µm thick) films as 0.7% and 0.1%, 136 respectively.

137 2.2. Buckling

138 Cotterell et al. [18,19] used the buckling test (cf. Fig. 139 4) to determine fracture toughness of indium-tin-oxide 140 (ITO) thin films. Polyethylene Telephthalate (PET) 141 polymer of a few tenth of millimeter in thickness was 142 used as the substrate due to its excellent elasticity. ITO 143 films with thickness between 80 and 140 nm were 144 deposited on the PET substrate. The testing scheme (cf. 145 Fig. 4) can be analyzed as a plane strain beam loaded 146 along its axis. According to large deformation buckling 147 theory of beams, the following equation can be obtained 148 [20]:

$$\chi = 2 \left[1 - \frac{E(k)}{K(k)} \right] \tag{4}$$

 $\frac{l}{R} = 4K(k)k$

150 where K(k) and E(k) are complete elliptic integrals, 152 $k=\sin(\theta/2)$, L is the original length of the beam, R is 153 radius of curvature, and $\chi=e/L$ contraction ratio. For the 154 two schemes in Fig. 4, l=L for simple support and l=L/2155 for built-in ends. The radius of bending can be calculated 156 by measuring the shortening of the beam e. Since the 157 ITO film is so thin compared with the substrate, the 158 neutral axis of the composite is very near to the center of composite. Therefore, the strain in the thin film can be 159 given by 160

$$\varepsilon = \frac{h_{\rm s} + h_{\rm f}}{2R} \tag{6}$$

where h_s and h_f are the thickness of the substrate and 162 film, respectively. Owing to the difference in conductivity 163 of the substrate and the film, cracking of the film can be 164 determined from a change in electrical conductivity. The 165 strain just before the sudden change in resistance is taken 166 as the critical strain ε_c , which is used to calculate the 167 critical strain energy release rate through [21] 168

$$G_C = \frac{1}{2} E_{\rm f} \varepsilon_{\rm c}^2 \pi h_{\rm f} g(\alpha, \beta) \tag{7}$$

where E_f is the elastic modulus of thin film, the factor $g(\alpha,\beta)$ 170 is a function of the Dundur's parameter, and the value of g 171 factor can be computed by finite element method [22]. Since 172 the PET polymer is used as the substrate, large elastic 173 deformation in substrate before film fracture becomes 174 possible. However, the low melting temperature of the 175 polymer substrate limits the application. 176

Scratch test is generally accepted as one of the simple 178means in assessing adhesion strength of a film on its 179substrate [23–25]. In the test process, a diamond tip is 180driven over a coated surface to produce a scratch. The load 181on the diamond tip is increased linearly to induce a shear 182force in the nearby film that is proportional to the applied 183 load and transmitted through the bulk of the composite 184sample. As the mechanical properties of the film and the 185substrate are different, there is a discontinuity in the shear 186stress at the interface which, when sufficiently high, induces 187adhesive failure at a critical load. Generally, for hard thin 188 films, microcracks appear in films during scratching before 189 the final adhesion failure [26]. The minimum load at which 190the first crack occurs is termed the lower critical load L_{c1} , 191 and the load corresponding to the complete peeling of the 192film is termed the higher critical load L_{c2} (cf. Fig. 5). Some 193researchers directly used the lower critical load to indicate 194cracking resistance [27,28], or some even termed it "scratch 195toughness" [29,30]. A magnetron sputtered $Ti_{1-x}Al_xN$ 196nanocomposite thin film (hardness 28.5 GPa, thickness 0.9 197µm) which reaches 70 N load without brittle failure at 198



(5)

Fig. 4. Schematic illustration of buckling test Left: simple support ends; right: built-in ends [19].

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Fig. 5. Scratch adhesion profile.

199 Rockwell diamond tip (200 µm in radius) is thus considered 200 having good scratch toughness. In comparison, films of TiN 201 (0.9 μ m, L_{c1} =30–40 N) and AlN (0.6 μ m, at L_{c1} =20 N) [31] 202 are not so good in scratch toughness. Since residual stress 203 affects critical load, multipass scratch test is proposed where 204 the scratch is performed in the same scratch track several 205 times with increasing load. Let the critical load L_{c}^{i} be the critical load after the *i*th pass; as the difference between L_c^{i} 206 207 and L_{c}^{i+n} ($n \ge 1$) becomes 0, the value of L_{c}^{i} is then taken as qualitative characterization of the "scratch toughness" [32]. 208209However, the critical load is not "fracture toughness" 210 (and, of course, the unit is wrong for fracture toughness!). 211 What the lower critical load represents is a load bearing 212 capacity or crack initiation load. Maybe it can be treated as 213 some sort of "crack initiation resistance": the higher the L_{c1} . 214 the more difficult it is to initiate a crack in the film. 215 However, initiation of a crack does not necessarily result in 216 fracture in the film; what is also important is how long the 217film can hold and withstand further loading before a 218catastrophic fracture occurs. Zhang et al. [33] pointed out 219that the film toughness should be proportional to both the 220 lower critical load and the difference of the higher and the 221lower critical load. The product of these two terms is termed 222"Scratch Crack Propagation Resistance," or CPRs:

$$CPR_{s} = L_{c1}(L_{c2} - L_{c1})$$
(8)

223 The parameter CPR_s can be used as quick qualitative 226 indication of the film toughness or used in a quality control 227 process for tough film. But CPR is not toughness.

Hoelm et al. [34] formulated an equation to related scratch test data to proper fracture toughness. The model assumes that cracking in the microscratch test is a result of cracks being opened on surface by the applied pressure at the bottom of the scratch groove. The coefficient of grooving friction can be calculated as the ratio of the tangential force (F) to normal force (P) (cf. Fig. 6). The stress intensity solution for a mode I crack opening is thus given as

$$K_{IC} = \frac{2Pf_{\rm g}}{R^2 \cot\theta} \left(\frac{a}{\pi}\right)^{1/2} \sin^{-1}\frac{R}{a} \tag{9}$$

where P is the pressure opening the crack, R is the radius of 230 238 the indenter cone into the groove, 2a is the total crack length, and f_g is the coefficient of grooving friction, which depends 239 on the cone angle 2θ and can be obtained from the track 240 width and the depth of penetration. However, this model is 241 oversimplified, and the actual state of forces in the groove 242 ahead and right below the tip are much more complicated 243 and have to be taken into account for better description of the 244 process. 245

More recently, Holmberg et al. [35] investigated the 246 fracture toughness of thin films through measuring of the 247 tensile stress, which induces the cohesion failure (i.e., 248 generation of cracks in the film) through 249

$$K = \sigma \sqrt{b} f(a, b) \tag{10}$$

where σ is the tensile stress which induces the cracks in film 250 during scratch obtained through a three-dimensional finite 252 element modeling, a is the crack length, and b is the 253 crack spacing, f(a,b) is a nondimensional function depend-254 ent on crack length a and crack spacing b measured from 255 the scratch track (cf. Fig. 7). As $a \gg b$, Eq. (10) reduces to . 256 Since the calculation of the tensile stress σ involves 3-D 257 finite element modeling and since a general f(a,b) expres-258 sion is not available, practical application of this method is 259 difficult. 260



Fig. 6. Schematic diagram of the microscratch fracture toughness measurement with a pressure P opening a crack (2a) out of a groove width 2R [34].



Fig. 7. Illustration of the scratch track and the cracks [35].

261 2.4. Indentation

262Perhaps indentation is the most widely used tool in assessment of thin film toughness. Plastic deformation leads 263264to stress relaxation in materials. The easier the stress relaxation proceeds, the larger plasticity is inherent in the 265material. Thus, comparing the plastic strain with the total 266strain in an indention test directly gives a simple, rough but 267quick indication of how "tough" the material is. Plasticity is 268269defined as the ratio of the plastic displacement over the total displacement in the load-displacement curve [36] (cf. Fig. 8). 270

$$Plasticity = \frac{\varepsilon_p}{\varepsilon} = \frac{OA}{OB}$$
(11)

where $\varepsilon_{\rm p}$ is the plastic deformation, and ε is the total 271 deformation. A superhard DLC film with hardness of 60 273 274 GPa has only 10% plasticity [37], whereas a "tough" nc-TiC/ a-C film with a hardness of 32 GPa has 40% plasticity 275 276 [30,38]. Hydrogen-free amorphous carbon films with hardness of 30 GPa has a toughness of 50% to 60% in plasticity 277 278 [39], depending on bias voltage during sputtering. Magnetron sputtered 1- μ m-thick Ti_{1-x}Al_xN films with hardness 31 GPa 279 280 obtained a plasticity of 32% [31].

However, "plasticity" is not fracture toughness. To measure a film's proper fracture toughness, Tsui et al. [40,41] introduced a precrack into the film using focused by ion beam milling. The crack opening force is generated by



Fig. 8. Schematic plot of a load-displacement curve obtained from nanoindentation. Plasticity is calculated as OA/OB.

means of indentation sink-in effect. The sink-in effect 285provides a tensile stress on the film near the precrack tip and 286promotes the crack propagation. A Knoop indenter is used 287to induce a plane strain condition near the indenter. The 288location of the indenter and precrack is schematically shown 289in Fig. 9. The extent of the sink-in effect and tensile stress 290generated at the precrack are the largest at the center of the 291indentation and decrease along the elongated edges of the 292indentation. Thus, the crack tip opening distance and crack 293growth are different at different locations along the 294precracked trench. Under the plain strain condition, fracture 295toughness can be expressed as a function of crack tip 296blunting immediately before the catastrophic failure through 297the following equation 298

$$K_{Ic} = \sqrt{m\delta\sigma_{\rm y}E} \tag{12}$$

where m is a dimensionless constant, approximately 2.0 for 300 a plane strain condition [40,42], $\sigma_{\rm v}$ is the yield stress 301 $(\sigma_v \approx H/3$ [2]), E is the Young's modulus of the films, δ is 302 the crack tip opening distance which is the amount of crack 303 tip blunting before the catastrophic crack growth. The 304 fracture toughness of NiP films with thickness of 9 µm 305 deposited on aluminum substrate is thus determined as 15 306 MPa $m^{1/2}$ [41]. The uncertainties associated with this 307 method come from the difficulty of making the precrack 308 and measurement of the crack tip opening distance. In 309 addition, for films on hard substrate, the sink-in effect may 310 not be induced that would render the method ineffective. 311

To avoid the difficulties in making the precrack, many 312 researchers directly indent the films without a precrack. 313 When the stress exceeds a critical value, a crack or 314 spallation will be generated. Failure of the film is manifested 315 by the formation of a kink or plateau in the load–displace- 316



Fig. 9. Schematic illustration of the orientation of the Knoop indentation relative to the precrack trench [40].

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Fig. 10. SEM observation of radial cracking at Vickers indentation.

317 ment curve or crack formation in the indent impression [43– 318 45]. As a qualitative, crude, and relative assessment, 319 Holleck and Schulz [46] compare the crack length under 320 the same load, and Kustas et al. [47] measures the "spall 321 diameter"—the damage zone around the indenter. More 322 quantitatively, length *c* of radial cracks (cf. Fig. 10) is 323 related to the fracture toughness (K_{IC}) through [48]:

$$K_{IC} = \delta \left(\frac{E}{H}\right)^{1/2} \left(\frac{P}{c^{3/2}}\right) \tag{13}$$

324 where P is the applied indentation load, E and H are the 326 elastic modulus and hardness of the film, respectively. δ is 327 an empirical constant which depends on the geometry of the 328 indenter. For standard Vickers diamond pyramid indenter and cube corner indenter, value of δ is taken as 0.016 [49] 329 330 and 0.0319 [50,51], respectively. The criterion for a welldefined crack is taken as $c \ge 2a$ [49], where a is the half of 331 332 the diagonal length of the indent. Both E and H can be 333 determined from an indentation test at a much smaller load 334 and analyzing of indentation load-displacement data [52]; 335 crack length c can be obtained using SEM, thus implementation of the method seems straightforward [53]. 336

337 However, the difficulty lies in the existence of a crack 338 formation threshold and locating and the determination of 339the crack length. Although indentation can be realized with 340 Vickers indenter, Berkovich indenter, or cube corner indenter [54,55], there exists a cracking threshold below 341342 which indentation cracking does not occur. Existence of the 343 cracking threshold causes severe restrictions on achievable 344 spatial resolution. The occurrence of the indentation cracking also depends on the condition of the indenter tip 345346 [56]. Harding et al. [57] found that indentation-cracking 347threshold could be significantly reduced by employing a 348 sharper indenter (cube corner indenter compared to the 349 Berkovich and Vickers indenters). The cube corner indenter 350 induces more than three times the indentation volume as 351compared to that by the Berkovich indenter at the same 352 load. Consequently, the crack formation is easier with the

cube corner indenter, thereby reducing the cracking thresh-353olds. For the cube corner indenter, the angle between the 354axis of symmetry and a face is 35.3° (as compared to 65.3° 355 for the Berkovich indenter), and there are three cracks lying 356in directions parallel to the indentation diagonal (cf. Fig. 357 11). Cracks that are well defined and symmetrical around 358the cube corner indentation are used to calculate the 359 toughness. Different researchers used different δ values: 360 0.0319 [50,51], 0.040 [57], and 0.0535 [58]. Despite the 361 inherited problems, due to its simplicity, the indentation 362method is widely used in toughness evaluation of thin films. 363 To cite a few: sputter-deposited DLC film (1.92 µm thick, 3641.57 MPa m^{1/2}) [59], plasma-sprayed Al₂O₃ (200-300 μm 365 thick, containing 13% TiO₂, 4.5 MPa m^{1/2}) [60], atmos-366 pheric pressure CVD SiC (3 µm, 0.78 MPa m^{1/2}) [61], 367 plasma-enhanced CVD nc-TiN/SixN (~1.5 µm, 1.3-2.4 MPa 368 m^{1/2}) [62], and TiC_xN_y/SiCN (2.7–3.3 μ m, ~1 MPa m^{1/2}) 369 [63]. 370

Some researchers suggest that the indentation load P, 371 radial crack length c, and fracture toughness K_{IC} have the 372 following relationship [64–66]: 373

$$\frac{P\chi_{\rm r}}{c^{3/2}} = K_{IC} - 2\sigma (c/\pi)^{1/2}$$
(14)

where σ is the residual stress at the surface. For a Berkovich 374 diamond indenter, $\chi_r = 0.016 (E/H)^{1/2}$. Since P, E, H, and c 376 are all experimentally attainable from the indentation test, plotting $P\chi_r/c^{3/2}$ against $2(c/\pi)^{1/2}$ yields a straight line, with 377 378 K_{IC} as the interception with the ordinate axis and the 379 residual stress as the slope. This method is developed for 380 bulk materials but has also been used to determine the 381 fracture toughness of an organic-inorganic hybrid coating of 382 3–20 µm in thickness on glass [66]. 383

The methods described above require measurement of 384indentation-induced radial cracks, which is usually possible 385 for relatively thick films. It could be difficult for thin and 386 ultrathin (≤ 100 nm) films. In the case of thin films, 387 indentation depth usually exceeds 10% of the film thickness 388to generate radial cracks. As such, the elastic-plastic zone 389 may already expand to the substrate [67,68]. Furthermore, 390because of the shallow indentation depths required in the 391indentation technique, it is often difficult to precisely 392 measure the radial crack length even under scanning 393 electron microscope (SEM) [69], presuming the indent 394



Fig. 11. Schematic diagram of median-radial crack systems for cube corner indentation.

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395 can be located after transferring from the indenter to the 396 SEM.

397 For thin films, a more reliable approach is the energy approach [70-74]: the energy difference before and after 398 cracking is considered responsible for fracture of the film, 399and the energy release rate, defined as the strain energy 400release per unit crack area, is calculated based on the energy 401 difference. Once the energy release rate is obtained, 402toughness is obtained from $\underline{K_c} = \sqrt{EG_c}$ (for plain stress 403mode I fracture), or $K_c = \sqrt{\frac{EG_c}{1-v^2}}$ (for plain strain mode I 404 405 fracture).

406As illustrated by Li et al. [69,70], fracture of hard films 407 under a load-controlled indentation measurement may be simplified into three stages (cf. Fig. 12): (1) the first ring-408 like through-thickness crack forms around the indenter by 409410high stress in the contact area; (2) delamination and buckling occur around contact area at the film/substrate 411 interface by high lateral pressure; (3) a second ring-like 412through-thickness crack forms, and spalling is generated 413 by high bending stresses at the edges of the buckled thin 414 films. 415

416 The strain energy released in the first/second ring-like 417 cracking and spalling can be calculated from the corre-418 sponding step in the load–displacement curve schematically 419 shown in Fig. 13. OACD is the loading curve, and DE is the 420 unloading curve. Area under curve ABC presents the energy 421 difference before and after the ring-like cracking, which is 422 released as strain energy to create the ring-like through-



Fig. 12. Schematic of the three stages in nanoindentation fracture in a film/ substrate system [69].



Fig. 13. Schematic diagram of a load-displacement curve showing a step during the loading cycle and associated energy release.

thickness crack. The fracture toughness of the film is then 423 written as 424

$$K_{IC} = \left(\frac{E}{\left(1 - v_{\rm f}^2\right)2\pi C_{\rm R}} \frac{\Delta U}{t}\right)^{1/2}$$
(15)

where E is the elastic modulus of the thin film, $v_{\rm f}$ the 426 Poisson's ratio of film, $2\pi C_R$ is the crack length in the film 427 plane, t is the film thickness, and ΔU is the strain energy 428 difference before and after cracking. In this method, the 429 cube corner indenter or the conical indenter is preferred 430 because the through-thickness cracking of thin films can be 431 accomplished at a low load, as demonstrated in a-C films 432 (0.1 µm in thickness) [75], TiN/Ti(C,N)/TiC multilayer of 433 total thickness of 8 µm [76], Ni/Al₂O₃ multilayers of 0.15 434 μm in thickness [77]. 435

A somewhat similar approach was illustrated by Toonder 436 et al. and Malzbender and de With [72,73] based on the 437 chipping (cf. Fig. 14) during indentation. The energy release 438



Fig. 14. The failure modes of chipping under Vickers indenter.

439 rate during chipping under Berkovich indenter is expressed 440 as

$$\Gamma = \frac{U_{\rm fr}^{\rm c}}{3\pi t' C_{\rm d}} \tag{16}$$

442 where C_d is the diameter of the delamination crack that 443 initiated the chip, t' the effective film thickness, which 444 accounts the fact that the crack does not propagate 445 perpendicular to the film/substrate interface. t' equals to 446 the film thickness t divided by $\sin(\delta)$, where δ is the average angle of the chipping edge. $U_{\rm fr}^{\rm c}$ is the energy dissipated 447 during the chipping, which can be determined from 448 analyzing the irreversibly dissipated energy (total energy 449 450 minus the elastic energy) $W_{\rm irr}$.

451 Ignoring the thermal energy, after one indentation cycle, 452 the irreversibly dissipated energy $W_{\rm irr}$ comprises the energy 453 dissipated due to the plastic deformation $(U_{\rm pl})$ and the 454 energy dissipated due to fracture or chipping $(U_{\rm fr})$, or

$$W_{\rm irr} = U_{\rm fr} + U_{\rm pl} \tag{17}$$

456 $W_{\rm irr}$ can be determined through computing the area between the loading and unloading curve. By plotting W_{irr} vs. the 457 indentation load, a curve can be obtained (cf. Fig. 15). The 458 459 onset of delamination changes the slope of the W_{irr} -P curve. Before chipping takes place, the total irreversibly dissipated 460 energy comprises that for plastic deformation (U_{pl}) and 461 delamination $(U_{\rm fr}^{\rm d})$. Upon chipping, $U_{\rm fr}$ comprises two 462 components: the energy release in delamination, $U_{\rm fr}^{\rm d}$ and 463 464 the energy released in chipping, $U_{\rm fr}^{\rm c}$, which is graphically attainable from the W_{irr} -P curve (Fig. 15). Plugging the 465 chipping energy back to Eq. (16) gives rise to the critical 466 strain energy release rate, which in turn yields fracture 467 468 toughness through Eq. (3).

469 Possible errors come from energy determination, crack 470 length, and area measurement due to irregularity in crack 471 shape. TiAlSiN thin films of hardness of 29–32.5 GPa and 472 thickness of 2 μ m have been measured this way, and a 473 toughness of 1.55–2.1 MPa m^{1/2} is reported [78]. Malz-



Fig. 15. Energy irreversibly dissipated during indentation as a function of the peak load applied during the indentation [70].

bender and de With used the SiO₂-filled methyltrimethoxysilane films (thickness of 2–4 μ m) to compare the measurements of fracture toughness by radial cracking method and the chipping method [79]. The results differ from 0% to 22%. 478

For relatively thick free-standing films, fracture toughness can be directly measured using tensile method 481 according to the ASTM standard E-399 [3], where the 482 precrack is introduced easily by laser cutting, as in the case of diamond films with thickness of 150–200 μ m [80]. The 484 fracture toughness of the films is thus measured as 5–6 MPa m^{1/2}, comparable with those from indentation methods. 480

For thin films, however, measurement without creating 487 the precrack makes more sense because of the difficulties 488 and uncertainties involved in making precracks in micron or 489submicron thin films. Harry et al. [81-83] propose a 490microtensile method in which a flat rectangular substrate 491of dimensions L (length) $\times w$ (width) $\times h_s$ (thickness) coated 492with a film of thickness $h_{\rm f}$ is put under tension. In the 493tension process, the cracking of the film causes energy 494 variations in the film/substrate system. For thin films 495 $(h_{\rm f} \ll h_{\rm s})$ coated on flat substrate $(h_{\rm s} \ll L)$ with perfect 496 adhesion (thus, buckling of the film will not happen), the 497film/substrate system can be regarded as a thin composite 498beam. The toughness of the film is calculated based on 499energy balance when the cracking occurs. 500

$$\Delta U = \frac{\left(\sigma_{\rm c}^{\rm f}\right)^2 (h_{\rm f})^2}{E^{\rm f}} \left\{ \pi f\left(\frac{E^{\rm f}}{E^{\rm s}}\right) + \frac{\sigma_{\rm c}^{\rm f}}{\sqrt{3}\sigma_{\rm y}^{\rm s}} \right\}$$
(18)

where ΔU is the net energy change, $f\left(\frac{E^{\rm f}}{E^{\rm s}}\right)$ is a function of 502 the elastic modulus ratio, and the values are tabulated in 503 Ref. [84] for different modulus ratios, $E^{\rm f}$ and $E^{\rm s}$ are the 504 Young's modulus for film and substrate, respectively. $\sigma_{\rm y}^{\rm s}$ is 505 the yield stress of the substrate. $\sigma_{\rm c}^{\rm f}$ is the effective critical 506 cracking stress which is experimentally determined. The 507 critical strain energy release rate for crack propagating 508 through the film $G_{\rm c}^{\rm f}$ is 509

$$G_{\rm c}^{\rm f} = \frac{\Delta U}{h_{\rm f}} = \frac{\left(\sigma_{\rm c}^{\rm f}\right)^2 (h_{\rm f})}{E^{\rm f}} \left\{ \pi f\left(\frac{E^{\rm f}}{E^{\rm s}}\right) + \frac{\sigma_{\rm c}^{\rm f}}{\sqrt{3}\sigma_{\rm y}^{\rm s}} \right\}$$
(19)

Fracture toughness of the film K_{IC} can be calculated **510** through Eq. (3) once G_c^{f} is obtained. Harry et al. [81] deposited W and W–C solid solution [W(C)] films of thickness from 1.8 to 16 µm on stainless steel substrates and subjected 515 the film/substrate composite beams under tension and measured the fracture toughness of the W films as 1.0 to 2.5 517 MPa m^{1/2} and that of the W(C) films as 0.2 to 1.0 MPa m^{1/2}.

The major drawback of this method lies in its requirement of substrate properties. Toughness of film is in fact a property of the film itself and should not vary with substrate. 521

Zhang et al. [85] propose a two-step uniaxial tensile method 522to characterize toughness of thin hard films. In this method, 523524 the film/substrate system is subjected to uniaxial tensile stress until the film fractures, while the substrate is still 525526elastic. After the loading is removed to allow complete elastic recovery, the system is subjected to a second loading 527until the previous extension. The onset of film fracture is 528determined by the loss of linearity in the load-extension 529curve. Upon failure, parallel cracks are generated in the film. 530531 The crack initiation and propagation patterns are examined using SEM, and crack density (number of crack per 532distance) is measured. The toughness of the thin film is 533then derived from the difference between the two subse-534535quent load-extension curves (cf. Fig. 16). Assuming that the 536film adheres perfectly to the substrate during loading and unloading, thus there is no interfacial cracking (i.e., no 537 adhesion failure); the elasticity of the substrate material is 538good enough for it to remain elastic while cracking occurs in 539the film; the aspect ratio of length to thickness of the 540541substrate is designed large enough to warrant a plain stress 542condition. Under the assumption of no adhesion failure, the 543energy variation ΔU in the film/substrate system is attributed to the through-thickness cracking in the film, 544545 which is the area difference of S_{OABCEO} - S_{ODEO} in Fig. 16.

$$\Delta U = \int_{\text{OABCEO}} P_1(x) dx - \int_{\text{ODEO}} P_2(x) dx$$
(20)

548 That can be experimentally determined and then related 549 to strain energy release rate. The energy release rate G_c is 550 defined as the strain energy release per unit crack area 551 [70,86]:

$$G_{\rm c} = \frac{1}{2} \left(\frac{1}{w\rho L f(\theta)} \right) \left(\frac{\mathrm{d}U}{\mathrm{d}C} \right) \tag{21}$$

552 where ρ is the crack density or number of cracks per unit 554 length (crack/µm) obtainable from the corresponding SEM 555 observations. *w* is the substrate width, *L* is the substrate 556 length, thus $w\rho L$ is the total crack length in the film plane,



Fig. 16. Schematic diagram of load-extension curve obtained using the two-step uniaxial tensile test under extension-control [85].

 $f(\theta)$ is a dimensionless factor dependent on crack orientation $(f(\theta) \ge 1)$. For cracks perpendicular to the film/substrate finterface, $f(\theta)=1$. For thin film, the through-thickness cracks propagate instantaneously, and the cracking is a single event, i.e., $dU/dC=\Delta U/h_{\rm f}$ [70], and, when the cracks are perpendicular to the interface, Eq. (21) is rewritten as 562

$$G_{\rm c} = \frac{1}{2} \left(\frac{1}{w\rho L} \right) \left(\frac{\Delta U}{h_{\rm f}} \right) \tag{22}$$

where h_f is the film thickness, $2w\rho Lh_f$ is the total crack area; 563 ΔU is the strain energy difference before and after cracking. 565 Since the film is in plain stress condition under Mode I 566 fracture, Eq. (3) holds. Plugging Eq. (22) into Eq. (3) gives 567 rise to toughness K_C as 568

$$K_C = \left[\left(\frac{E}{2w\rho L} \right) \left(\frac{\Delta U}{h_{\rm f}} \right) \right]^{1/2} \tag{23}$$

where *E* is the Young's modulus of the thin film. A case 569 study of hard nanocomposite nc-TiN/a-SiN_x films of 3.0 μ m 571 thick gives a toughness value of 2.6 MPa m^{1/2} [85]. 572

The advantage of this two-step tensile method lies in its 573 independence from substrate properties, the ease, and speed 574in experimentation. The principle of the method, data 575treatment, and sample preparations are simple. In addition, 576the tensile test covers more area, and the property thus 577 characterized is more close to the material intrinsic property 578compared with the indentation or bending methods. The 579drawback of the method is the elasticity requirement of the 580substrate: the substrate has to remain in elastic deformation, 581while the coated film has undergone fracture. The second 582important requirement is the perfect adhesion between film 583 and substrate, without which significant variation in experi-584ment results may occur. 585

3. Summary and ending remarks 586

Toughness measurement for thin films is difficult due to the small dimension in thickness. Until now, there is neither standard procedure nor standard methodology. As in qualitative or semiquantitative assessment, sometimes a quick plasticity measurement or a scratch adhesion test (for crack propagation resistance) will suffice. But these test results should not be termed "toughness." 593

More elaborate quantitative measurements can be cate-594gorized into two main groups: the stress approach and the 595energy approach. The stress approach examines the stress 596state near the tip of the crack. Toughness is obtained through 597 $K_C \alpha \sigma_f \cdot \sqrt{a}$. Bending with precrack, scratch in consideration 598 of critical tensile stress, crack length and spacing, and 599 indentation in consideration of the critical load and 600 corresponding crack length, etc., fall into this category. 601 Difficulties in these methods lie in the formation of the 602 precrack, determination of the crack length, and the critical 603 10

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604 failure stress. These problems are not easily resolved due to 605 the thickness dimension involved.

606 The energy approach concentrates on the system's energy 607 state before and after fracture of the film. This energy 608 difference is considered consumed to increase new crack 609 area. Toughness is thus obtained through the critical energy 610 release rate $G_c: K_c = \sqrt{EG_c}$ (for plain stress mode I 611 fracture) or $K_C = \sqrt{\frac{EG_c}{1-v^2}}$ (for plain strain mode I fracture), 612 where G_c is related to the energy difference (ΔU) and crack 613 area (ΔA) through $G_c \Delta U / \Delta A$. Fallen under this category are 614 bending without precrack in the film, buckling, indentation 615 with chipping, tensile tests, and so on.

Standardization of thin film toughness measurements 616617 seems necessary. Energy-based methodologies have clear 618 advantages over stress-based approach.

619 In order not to confuse with the classical concept of 620 fracture toughness, it is strongly suggested that the term 621 "fracture toughness" not be used in thin film toughness

622 description where precrack is not involved. Instead, simply, "toughness" should suffice. 623

624 Acknowledgement

625 This work was supported by Nanyang Technical Uni-626 versity's research grant RG12/02.

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